

Design and Evaluation of Enhanced Dielectric-Barrier-Discharge Actuators

Stephen Wilkinson, NASA LaRC, Flow Physics and Control Branch

Emilie Siochi, NASA LaRC, Advanced Materials Processing Branch

Godfrey Sauti and **Tian-Bing Xu**, National Institute for Aerospace (NIA)

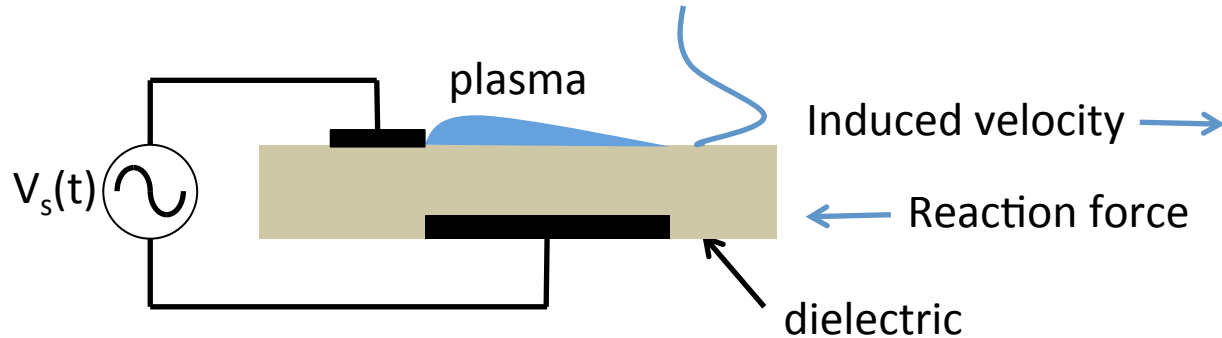
Mary Ann Meador, GRC, Durability and Coatings Branch

Hiaquan Guo, Ohio Aerospace Institute (OAI)



Dielectric Barrier Discharge Actuator

NASA Aeronautics Research Institute



Non-thermal, non-equilibrium plasma



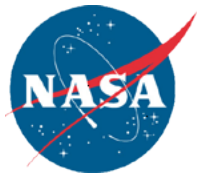
Typical Dielectric Barrier Discharge on Glass
 $f=5$ kHz



Outline

NASA Aeronautics Research Institute

- Relevance to NASA Aeronautics
- Description of effort's scope and rationale
- Experimental effort (dielectric and DBD force data)
- Alternative 1-D volume discharges
- Conclusions



Can DBD Actuators Support High Reynolds Number Flight Flow Control?

NASA Aeronautics Research Institute



- While electronic DBD flow control is currently a relatively weak effect, its robustness, controllability and simplicity could make it valuable option for future generations of aircraft.
- In order to fulfill that vision, pursuit of new innovations and understanding physical mechanisms must proceed.



Leveraging Activities

NASA Aeronautics Research Institute

- **NASA Aeronautics Research Institute (NARI) Seedling Award**
 - “Enhanced Dielectric-Barrier-Discharge Body-Force Generation Using Nanofoam Materials” – Phases 1 & 2
 - NARI seeks and develops new and promising aeronautics ideas for transfer to mainline project funding.
 - Current Goal: Explore materials of varying microstructure and chemistry, particularly **aerogels/nanofoams**, for enhanced performance of DBD flow control actuators.
- **NATO STO/CSO AVT-190 – “Plasma Based Flow Control for Performance and Control of Military Vehicles”**
 - Goal: Provide DBD code validation data (thrust, velocity) agreed upon by multiple NATO and PfP (Partners for Peace) participants.



Dielectric Terminology

NASA Aeronautics Research Institute

Poisson's equation drives DBD electrostatic forces

$$\bar{\nabla} \cdot \bar{D} = \rho_c$$

Definition of displacement field

$$\bar{D} = \epsilon_o \bar{E} + \bar{P} = \epsilon_o \epsilon_r \bar{E}$$

P = polarization field

D = displacement field

E = electric field

ϵ_o = permittivity of free space

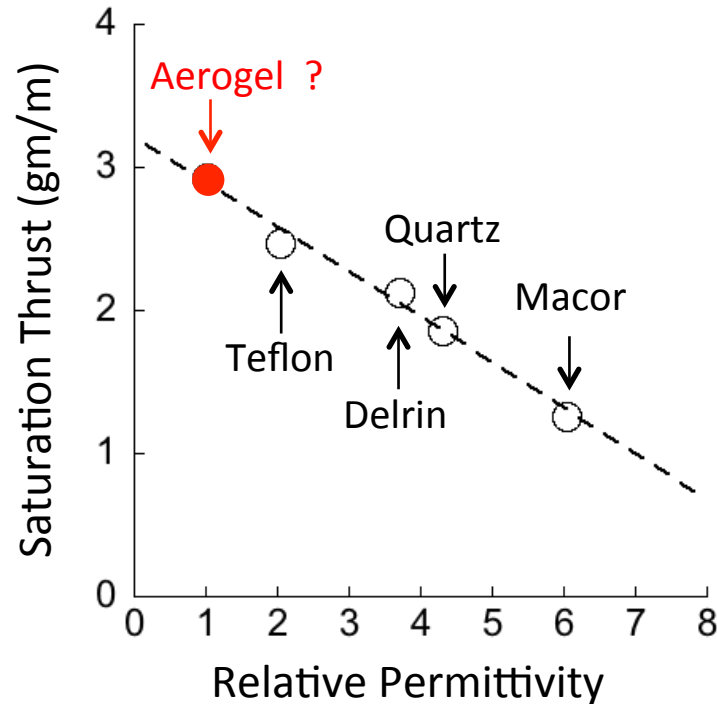
ϵ_r = relative permittivity of material

ρ_c = charge density



Force Enhancement with Low Permittivity Dielectrics

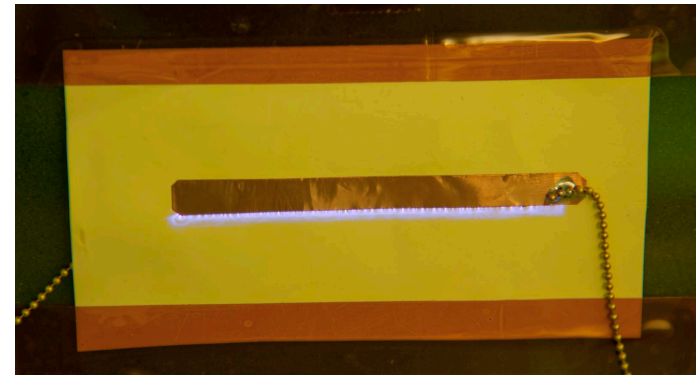
NASA Aeronautics Research Institute



Adapted from :
Thomas F., et. al., AIAA J., 2009 and
Durscher, R. and Roy, S., J. Phys. D: Appl. Phys., 2012

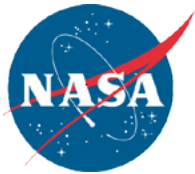


NASA/GRC Polyimide Aerogel*



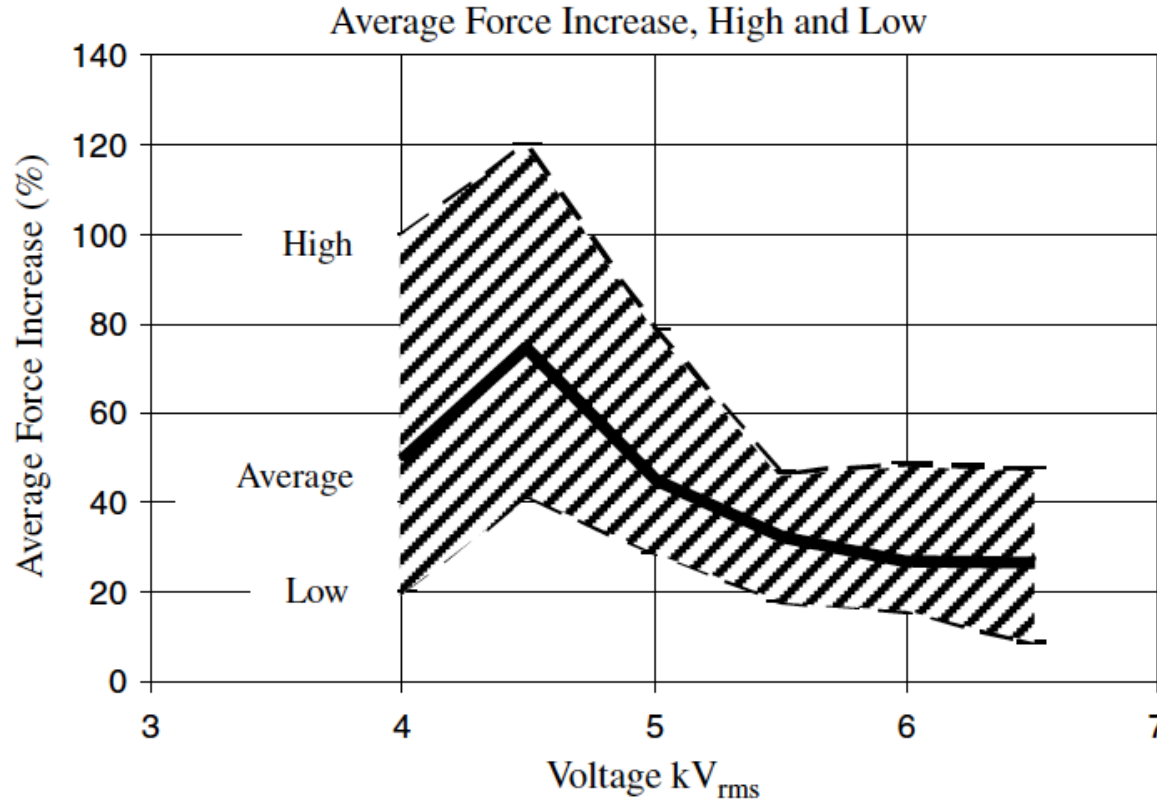
Aerogel Supports DBD Plasma

*<https://technology.grc.nasa.gov/featured-tech/aerogels.shtm>



Force Enhancement with Surface Coatings

NASA Aeronautics Research Institute



This may be a humidity effect and not a catalytic phenomena

Titanium Oxide: Photocatalyst and Wide Band-Gap Semiconductor)

From: Fine and Brickner, AIAA J., No. 12, Dec 2010



Guiding Questions

NASA Aeronautics Research Institute

- Are some materials superior to others in their ability to mediate creation of DBD body force?
- What is the role of microstructure and chemical composition?
- How do volume and surface properties affect performance?
 - Permittivity and dissipation
 - Secondary electron emission (SEE)
 - Catalytic/photocatalytic effects
 - Surface conductivity and chemical coatings
 - Surfactants (anionic, cationic, non-ionic)
 - Antioxidants (radical scavenger)
 - Adsorbed moisture (relative humidity)



Goals

NASA Aeronautics Research Institute

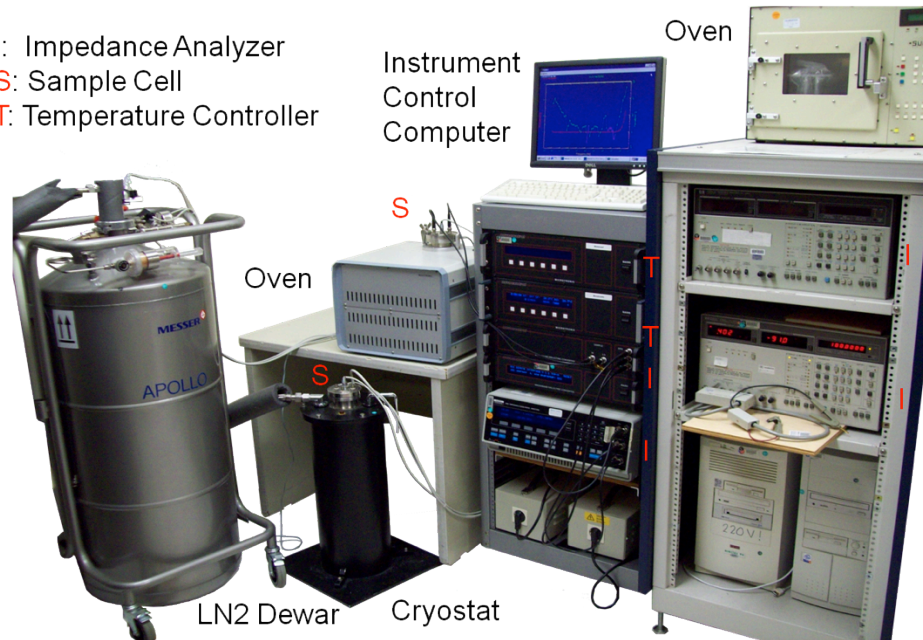
1. Characterize a variety of existing dielectric materials seeking to explain differences in body force generation using:
 - Dielectric properties
 - Reaction force
 - Electrical charge transfer
 - Lumped-parameter 1-D circuit simulations
2. Use this information to develop new materials optimized for DBD applications



Equipment for Materials Characterization

NASA Aeronautics Research Institute

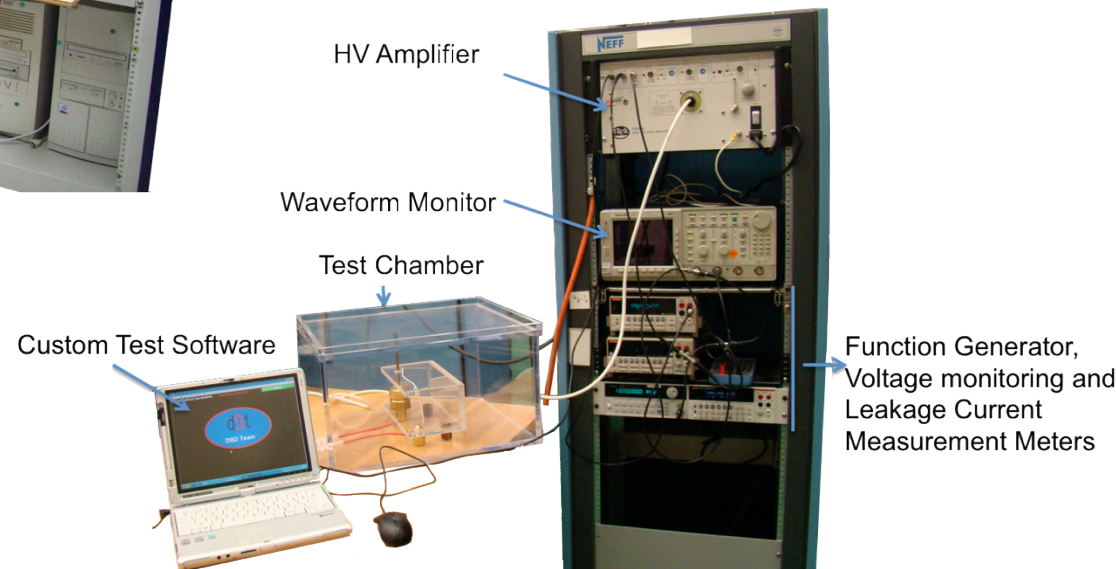
I: Impedance Analyzer
S: Sample Cell
T: Temperature Controller



Dielectric Breakdown
Strength



Dielectric Constant
and Loss Tangent

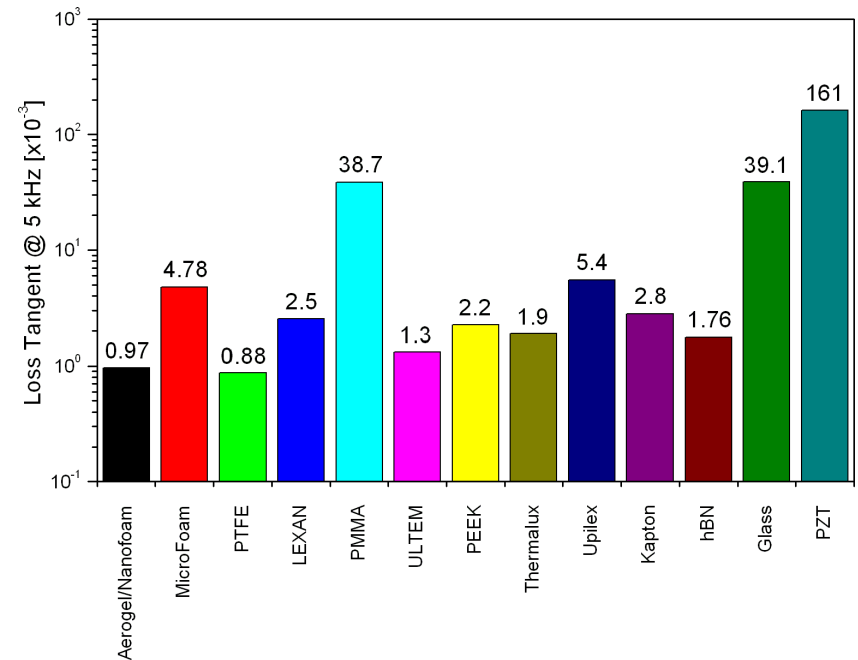
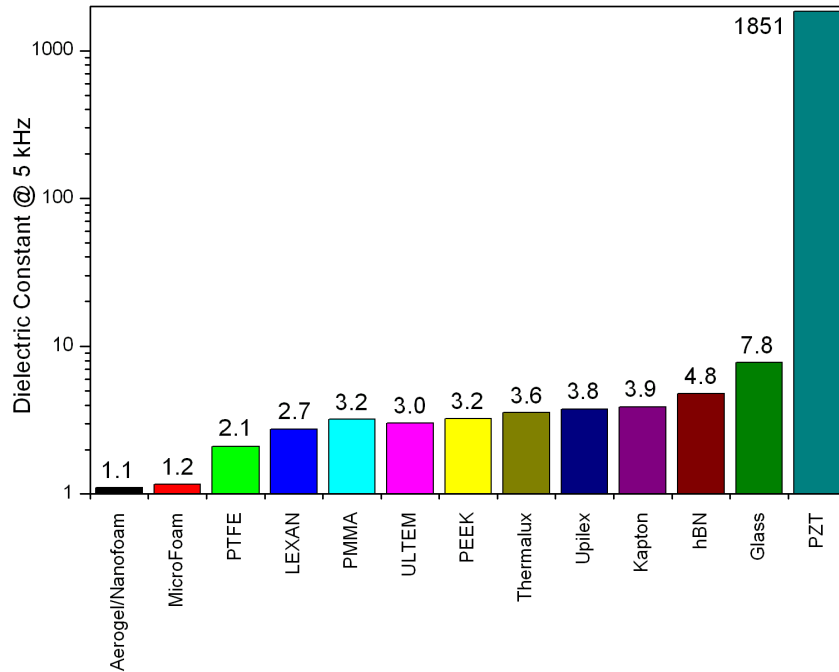




Measured Dielectric Constant for Various Materials

NASA Aeronautics Research Institute

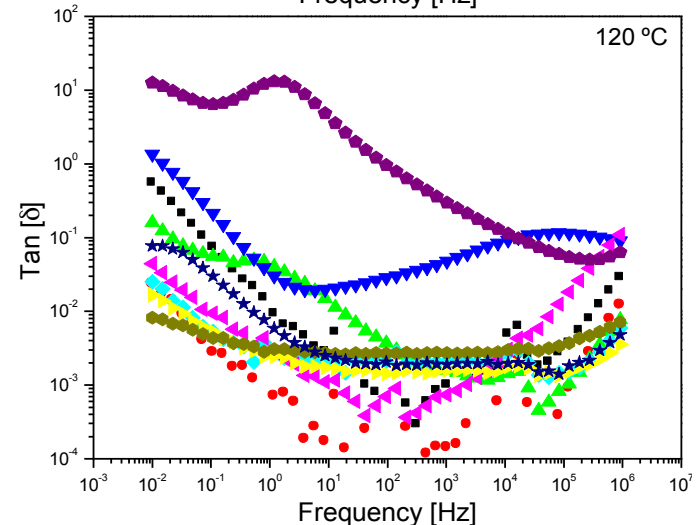
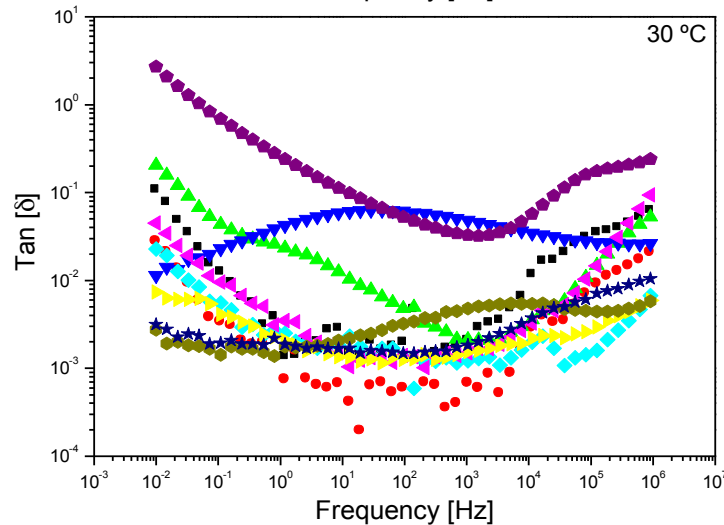
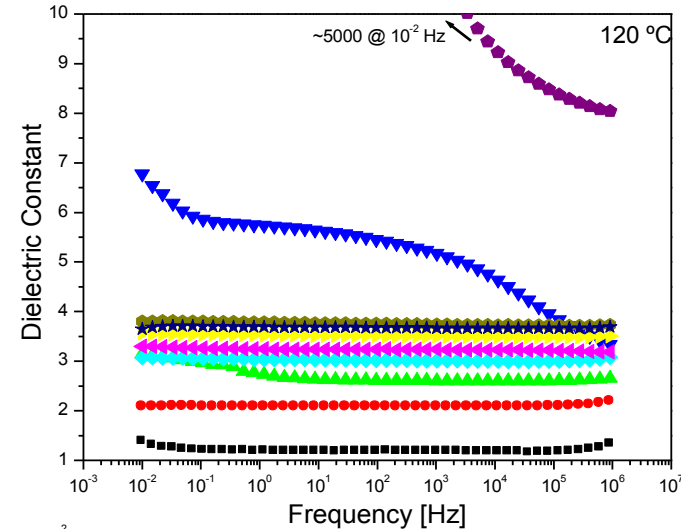
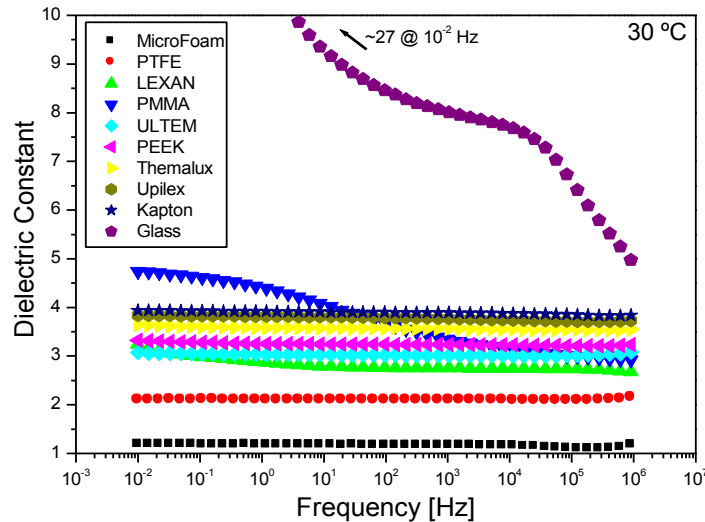
Desired properties:
Low dielectric constant and loss





Measured Dielectric Constant for Various Materials

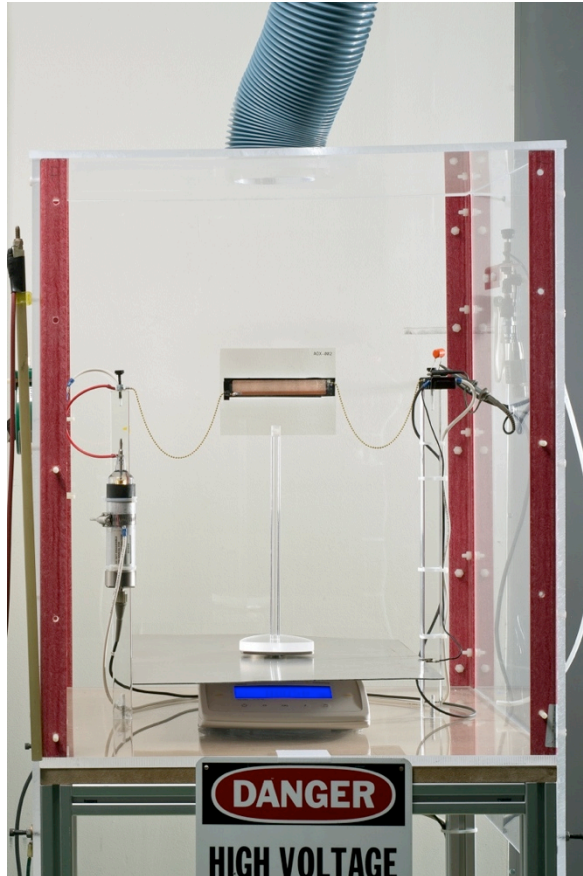
NASA Aeronautics Research Institute



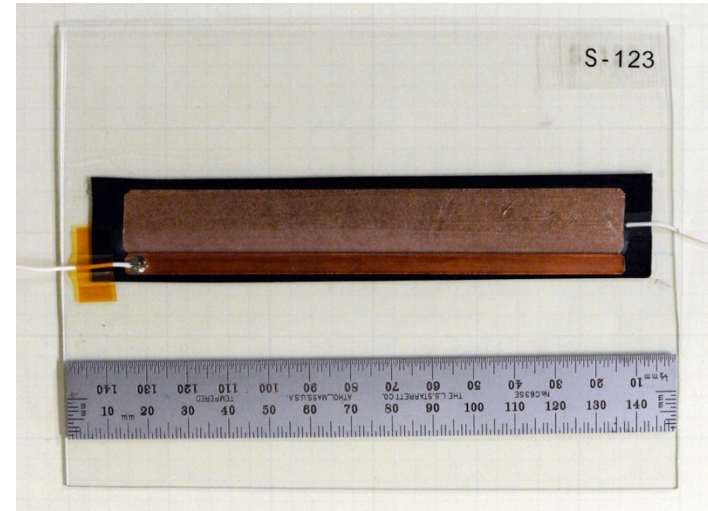


DBD Thrust Measurement Equipment

NASA Aeronautics Research Institute



FPCB DBD Thrust Stand



Typical DBD Model

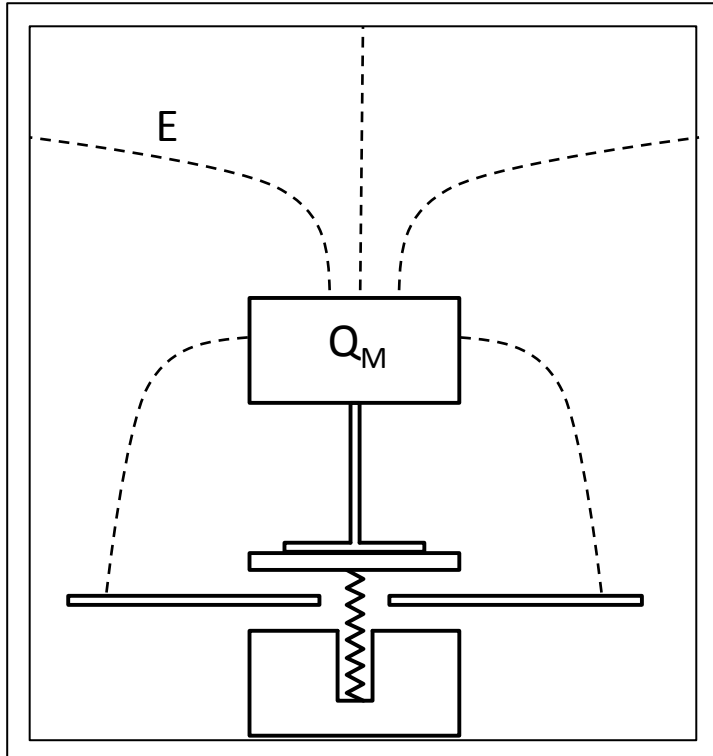


Original Thrust Stand, USAFA, 2005



Error Due to Electrostatic Induction Between Model and Surroundings

NASA Aeronautics Research Institute



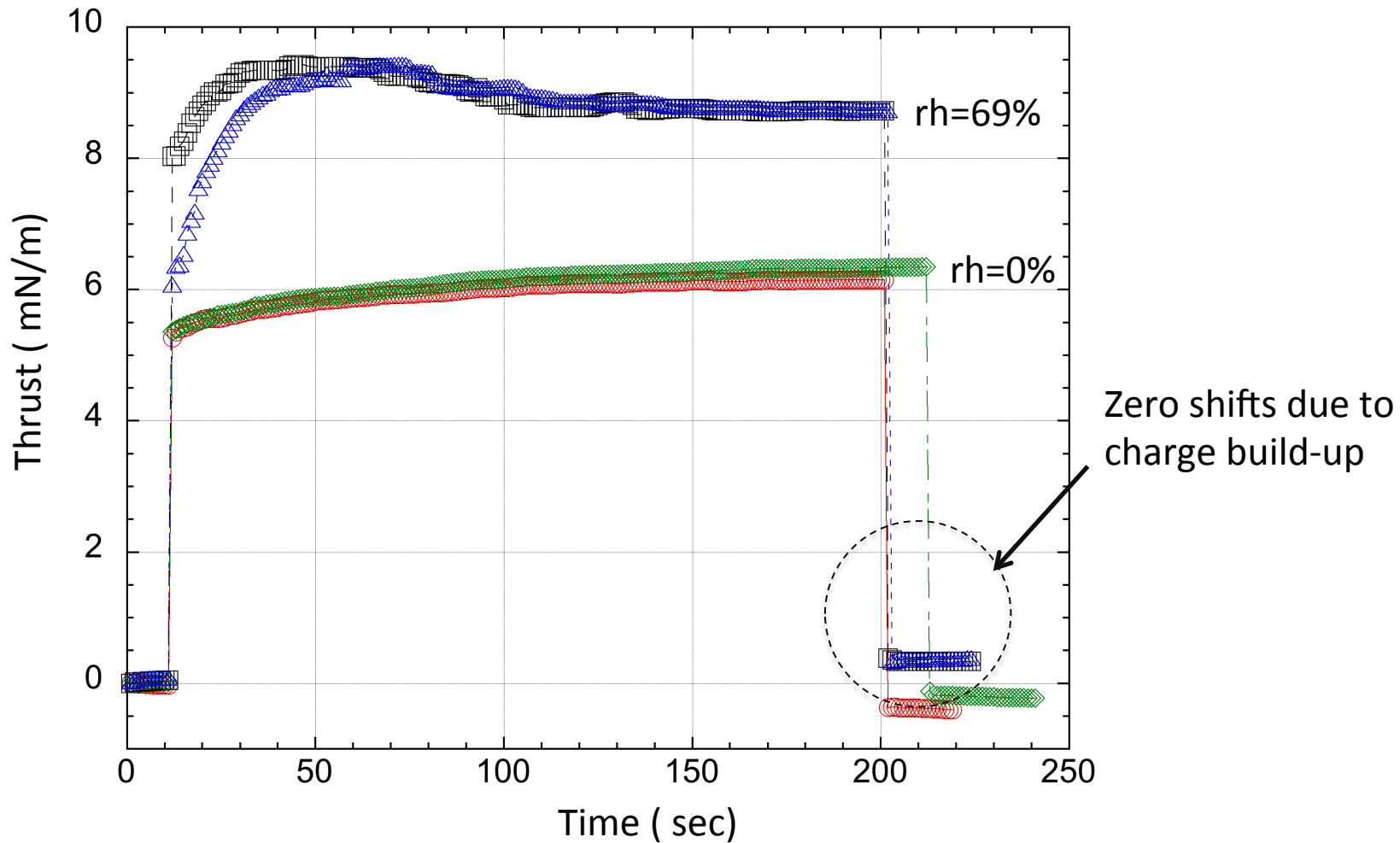
Force balance and shield grounded

- Model with charge Q_M is in electrostatic force equilibrium with surroundings.
- Uncontrolled physical or electrical (charge) changes may lead to measurement error.
- Humidity in the enclosure has a large, first order effect on DBD force.



Effect of Relative Humidity: Clean Glass Substrate

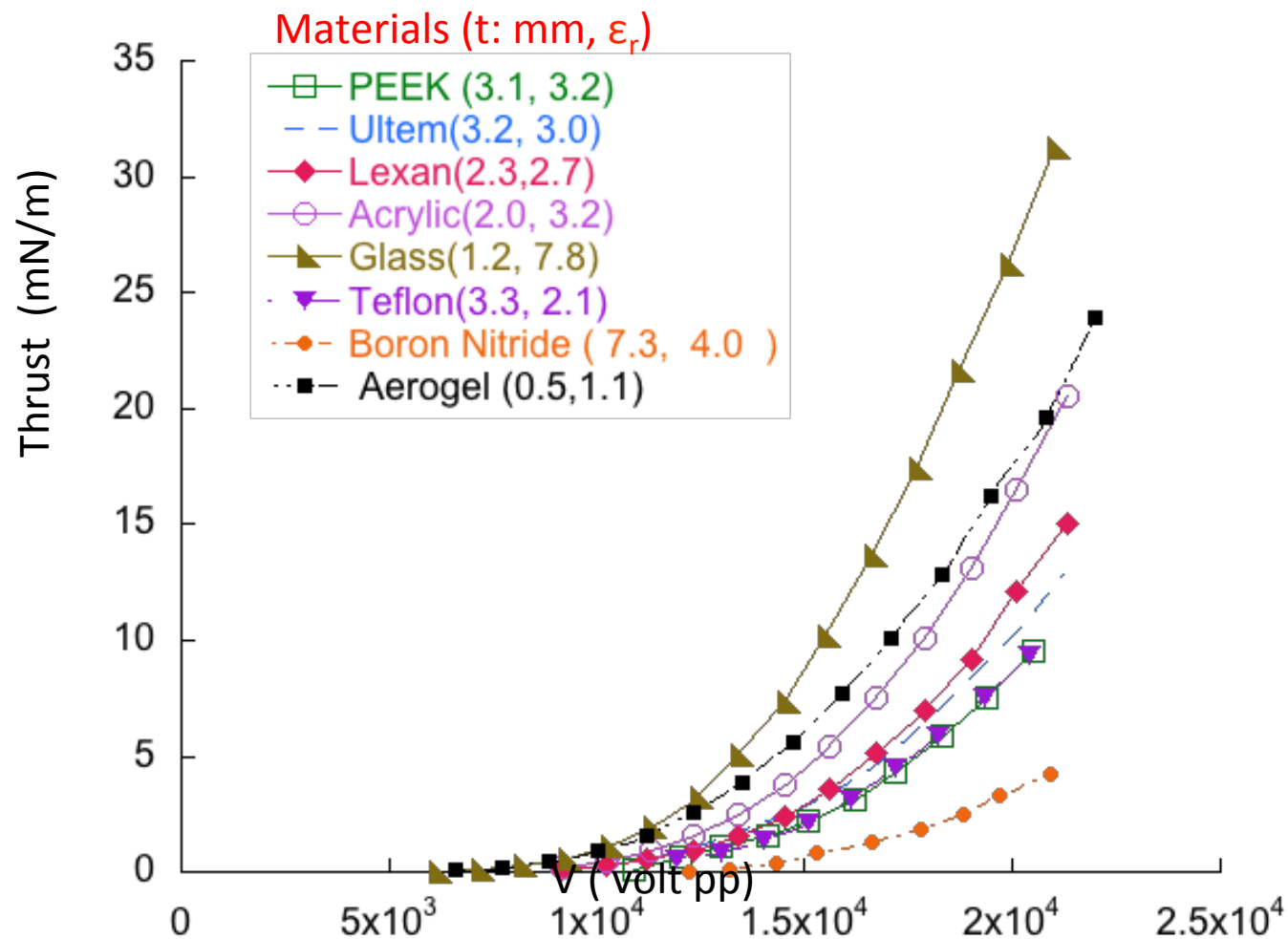
NASA Aeronautics Research Institute

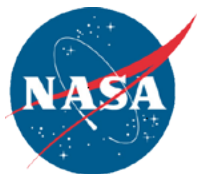




Thrust Data for Dielectric Materials

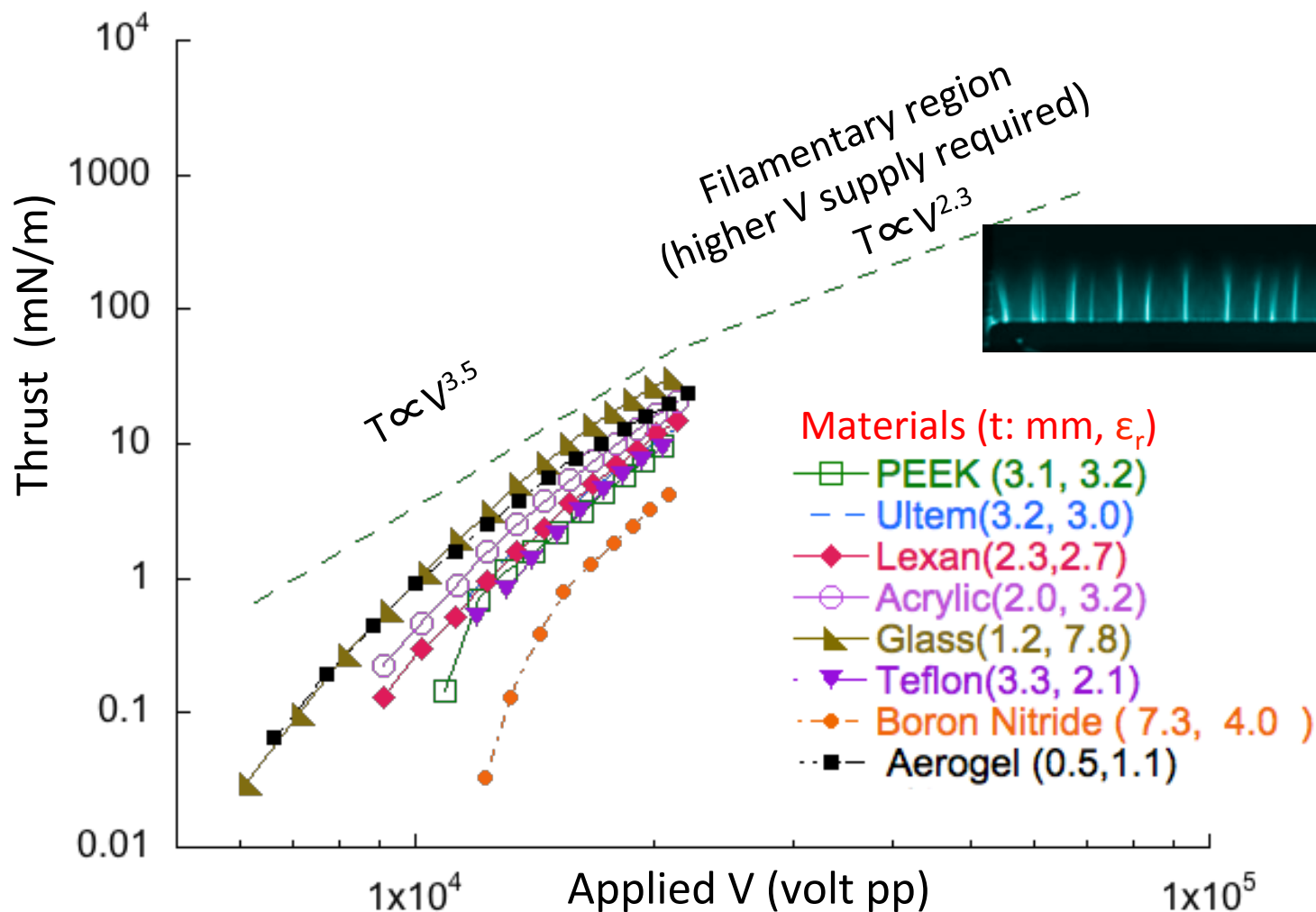
NASA Aeronautics Research Institute





Thrust Data for Dielectric Materials

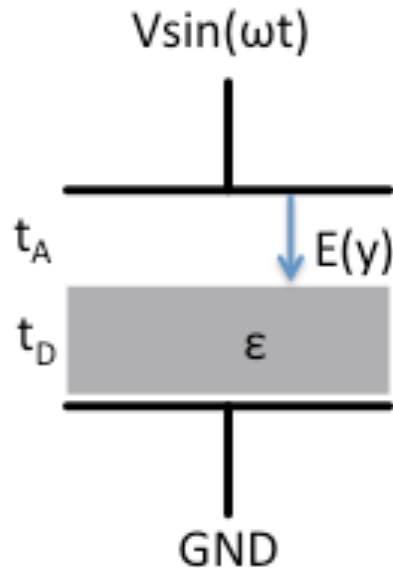
NASA Aeronautics Research Institute



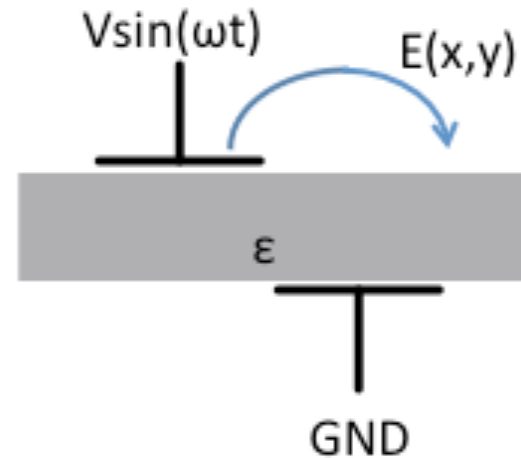


1D Volume Discharge May Offer Advantages for Dielectric Testing

NASA Aeronautics Research Institute



Volume Discharge
(One-dimensional)

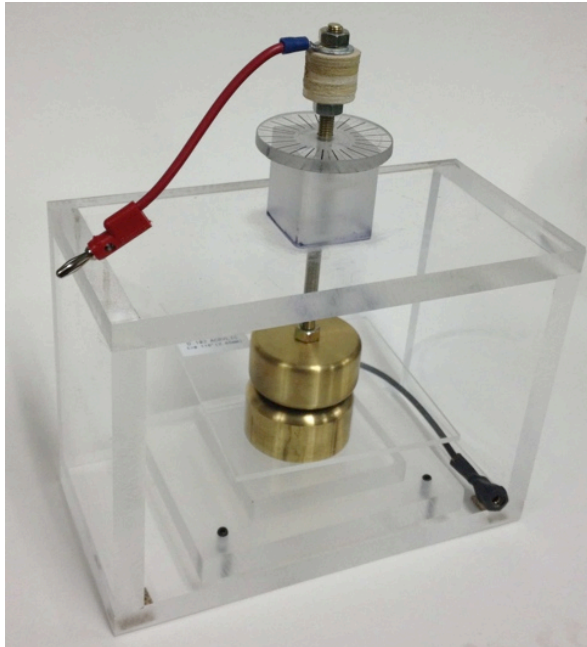


Surface Discharge
(Two-dimensional)

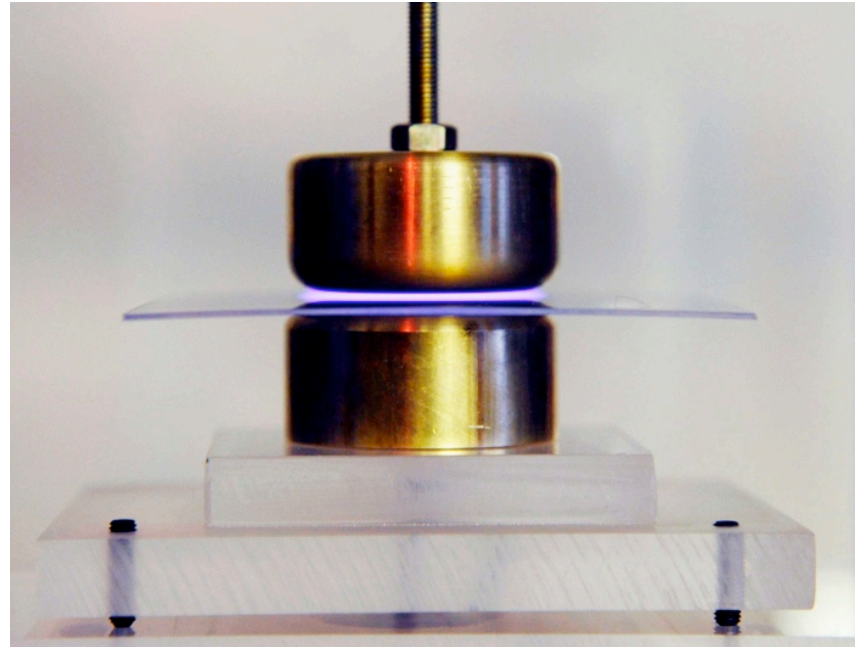


One Dimensional Volume Discharge Apparatus

NASA Aeronautics Research Institute



Variable gap volume
discharge tester



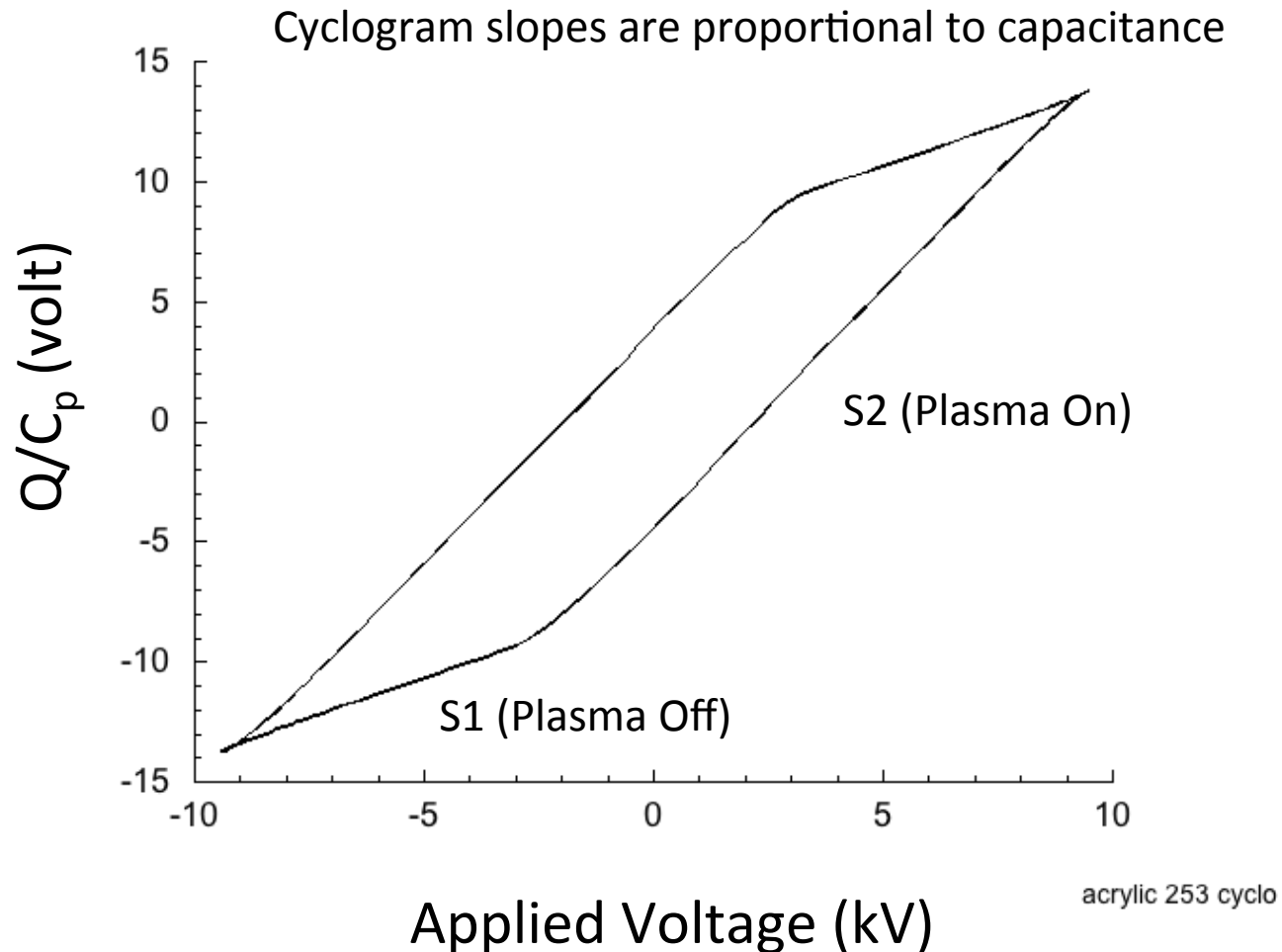
Typical DBD volume discharge

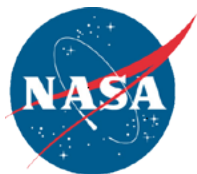


Typical 1-D Cyclogram

Acrylic, thickness=0.94 mm, $\epsilon_r=3.2$, $f=1$ kHz

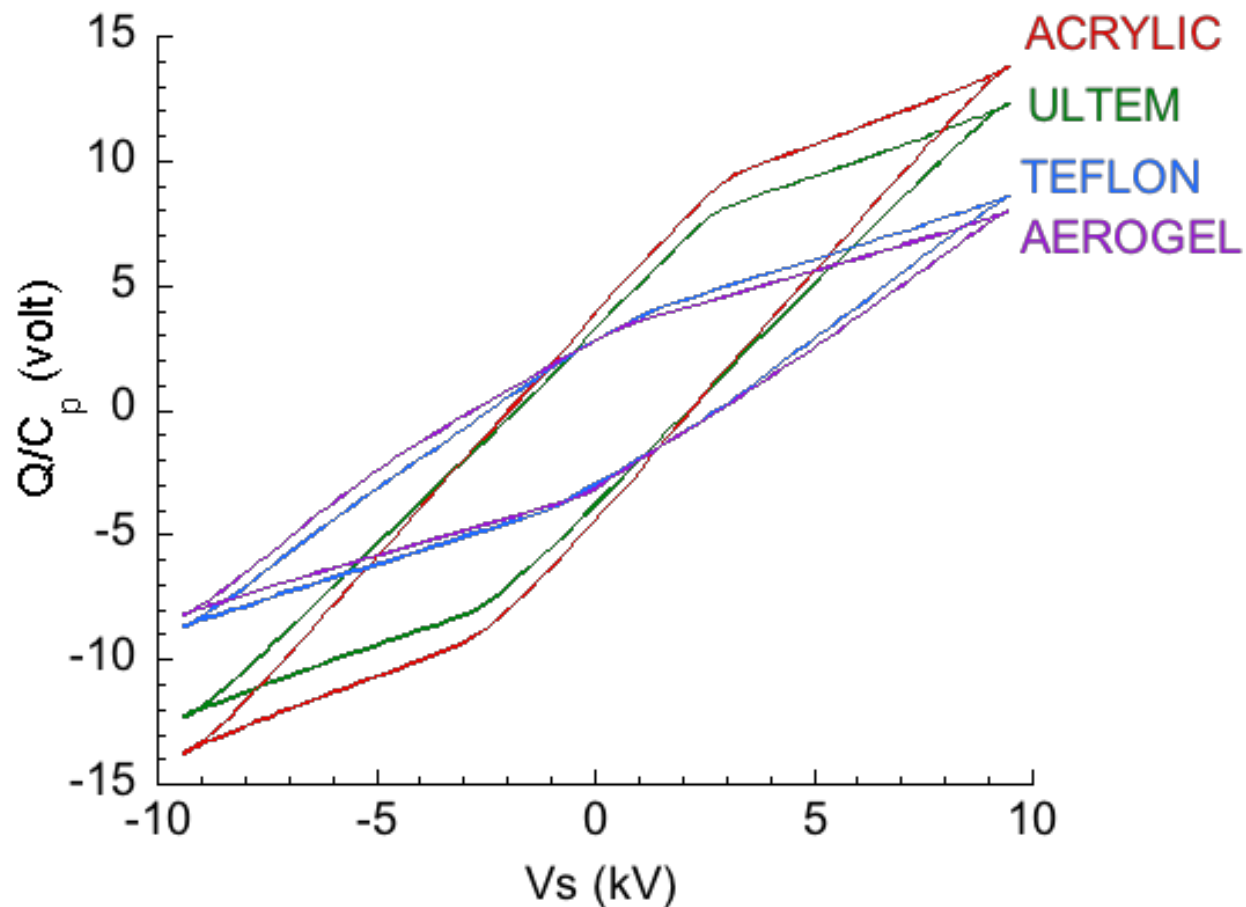
NASA Aeronautics Research Institute

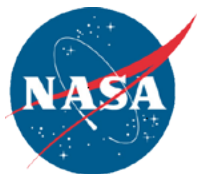




Cyclogram Characteristics for different materials

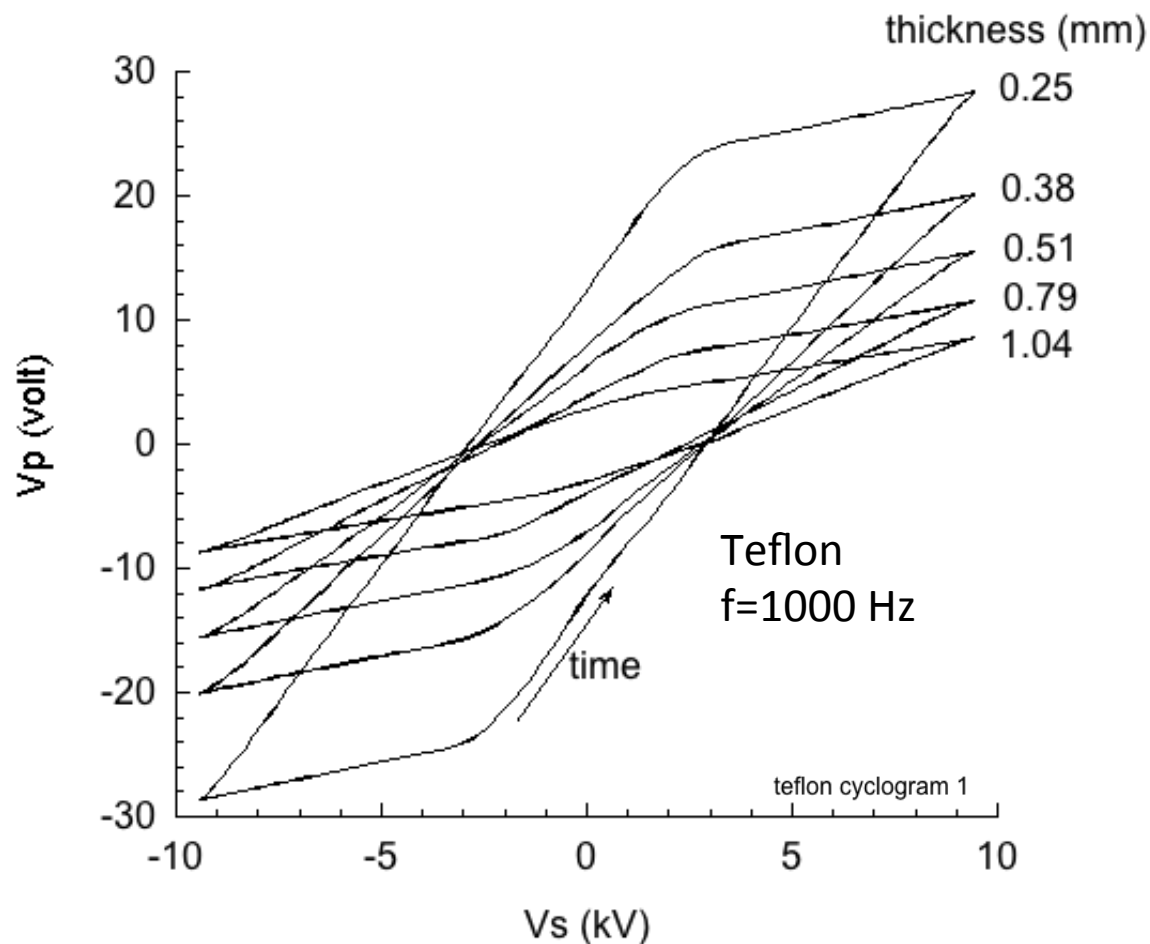
NASA Aeronautics Research Institute

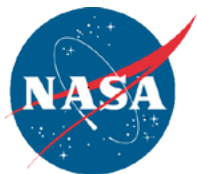




Parametric Analysis of Cyclogram Characteristics (frequency, voltage, gap, thickness, permittivity)

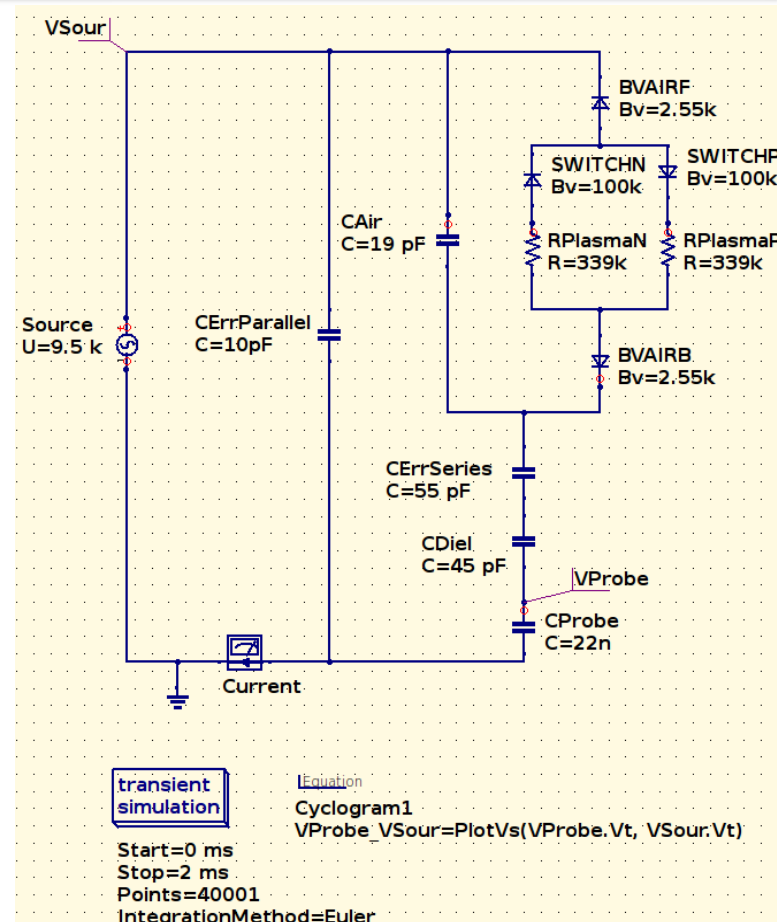
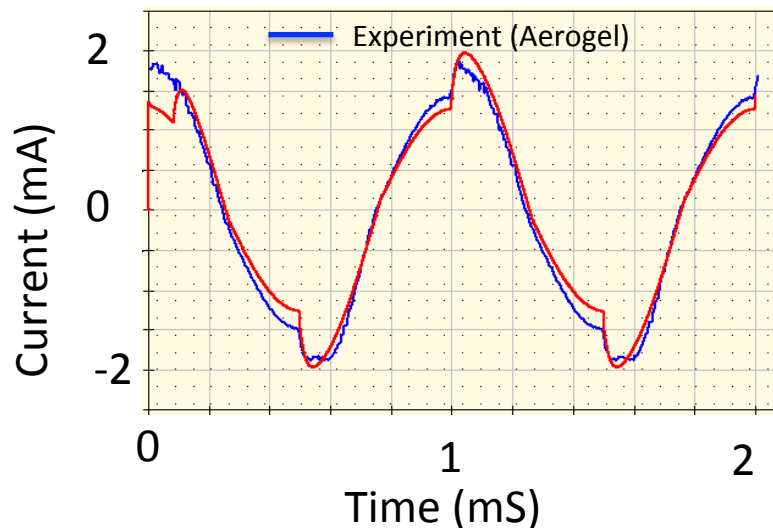
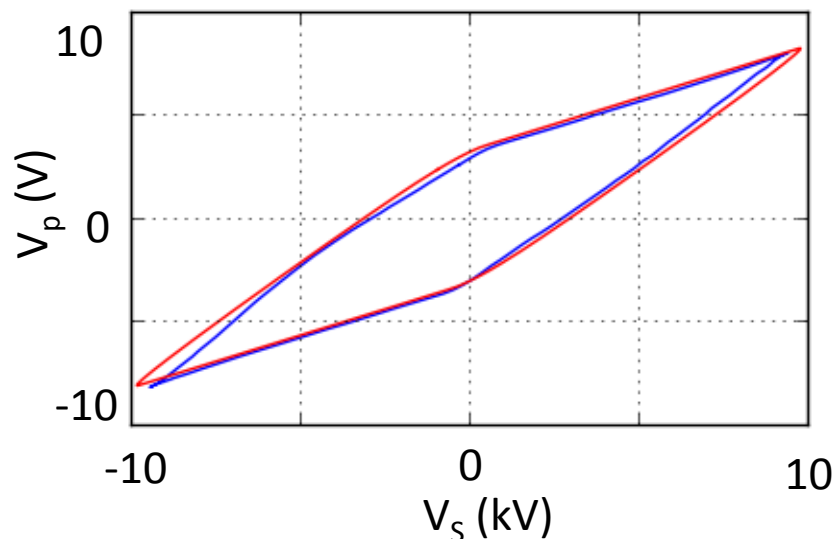
NASA Aeronautics Research Institute



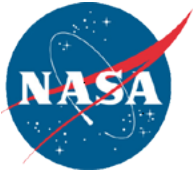


One Dimensional Circuit Simulation of the Actuator Material and Test Environment*

NASA Aeronautics Research Institute



*Data fitting using Qucsator (0.0.18) and SageMath (5.12)



Summary / Conclusions

NASA Aeronautics Research Institute

- Goal 1: Characterize existing dielectric materials seeking to explain differences in body force generation.
 - Implemented dielectric and force diagnostics, surface coatings, circuit simulations and 1D charge transfer technique.
 - Studied a dielectric properties and performance of a variety of organic and inorganic materials.
 - Demonstrated that polyimide aerogels can support a DBD plasma.
 - Measurement uncertainties due to inadequately controlled system electrostatics and variable atmospheric humidity preclude conclusive material assessments at this time.



Summary / Conclusions (cont.)

NASA Aeronautics Research Institute

- Goal 2: Use this information to develop new materials optimized for DBD applications.
 - Refined diagnostics with increased resolution and accuracy are required to differentiate DBD responses due solely to bulk permittivity and geometry from those due to possible plasma/surface (material) interaction effects.
 - Higher voltage power supply (> 20kV pp) is required to examine force saturation effects where surface interactions are more likely to be manifest.



Future Directions

NASA Aeronautics Research Institute

- Continue fundamental physics research of DBD dielectric materials.
- Identify specific flow features amenable to active DBD flow control and align DBD research with aeronautical applications
- Provide estimates of the impact of DBD forcing on airplane system energy resources.
- Identify appropriate program(s) within ARMD best suited for DBD research and other related direct, ionized gas flow control concepts (plasma arcs, spark jets, etc.)



And if all else fails: Consider *DBD Art and Music* !

NASA Aeronautics Research Institute



Copper foil on 1.2 mm thick glass
 $V = 20 \text{ kVpp}$ @ 1 kHz
Nikkor 50mm f/1.4 , 1/5 sec exp.

Concept: Ken Miller, Eagle Harbor Technologies, Seattle, WA



Tone scale (C_6)
 $V = 20 \text{ kVpp}$
 $1046.5 < f < 2093.0$



END